



# National Transportation Safety Board

## Aviation Incident Data Summary

<b>Location:</b>	Midland, TX	<b>Incident Number:</b>	FTW021A027
<b>Date &amp; Time:</b>	11/02/2001, 0734 CST	<b>Registration:</b>	N814AW
<b>Aircraft:</b>	Airbus Industrie A319-132	<b>Injuries:</b>	89 None
<b>Flight Conducted Under:</b>	Part 121: Air Carrier - Scheduled		

### Analysis

The transport airplane was in cruise flight at flight level 390 for 12 minutes when they received an "engine oil filter bypass" fault message on the engine centralized aircraft monitoring (ECAM) system; however, all of the engine parameters remained within limits. Subsequently, the oil pressure indication for the #1 (left) engine rose into the red band and a "high vibration and a thumping sound" was felt and heard. The flight crew then declared an emergency and diverted to another airport. The captain reported that during the landing roll, he moved both throttle levers into reverse, and simultaneously the cockpit and cabin began to fill with smoke. Air traffic controllers reported they observed white smoke emanating from the #1 engine during the landing roll. The captain stopped the airplane on the high-speed taxiway, turned off both engines, and an emergency evacuation ensued. The 1L and 2L doors were operated normally; however, the 1R door jammed when the flight attendant attempted to open it. Examination of the 1R door actuator and slide did not reveal the reason it failed to operate. Examination of the engine revealed that debris contamination of the #3 bearing initiated spallation of the bearing's outer ring raceway. Cyclic loading from the bearing balls passing over the growing raceway spall resulted in extensive fretting of the outer diameter surface of the outer ring, from which a fatigue crack was initiated. High-cycle fatigue progression radially through the outer ring was followed by rapid fracture and subsequent liberation of the outer ring fragments. The debris contamination more than likely came from the high-pressure compressor (HPC) stubshaft coating, which was liberated and entered the #3 bearing area causing it to fracture, and the engine to lose power. Research revealed this was one of five similar occurrences, which was traced down to a change in the manufacture process for the HPC stubshaft coating. The manufacturer has taken actions to alert operators of the existing problem.

### Probable Cause

The National Transportation Safety Board determines the probable cause(s) of this incident to be: the #1 engine's fatigue failure of the #3 bearing due to the manufacturer's inadequate design of the high-pressure compressor stubshaft coating, which resulted in a loss of engine power and an emergency landing.

### Findings

Occurrence #1: LOSS OF ENGINE POWER(PARTIAL) - MECH FAILURE/MALF  
Phase of Operation: CRUISE - NORMAL

#### Findings

1. 1 ENGINE
2. (C) TURBINE ASSEMBLY, SHAFT BEARING - FATIGUE
3. (C) MATERIAL DEFECT (INADEQUATE QUALITY OF MATERIAL) - MANUFACTURER

#### 4. (C) TURBINE ASSEMBLY, SHAFT BEARING - FAILURE, TOTAL

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Occurrence #2: FORCED LANDING

Phase of Operation: DESCENT - EMERGENCY

Findings

5. PRECAUTIONARY LANDING - PERFORMED - FLIGHTCREW

### Pilot Information

<b>Certificate:</b>	Airline Transport; Flight Instructor; Commercial	<b>Age:</b>	45
<b>Airplane Rating(s):</b>	Multi-engine Land; Single-engine Land	<b>Instrument Rating(s):</b>	Airplane
<b>Other Aircraft Rating(s):</b>	None	<b>Instructor Rating(s):</b>	Airplane Single-engine
<b>Flight Time:</b>	15000 hours (Total, all aircraft), 800 hours (Total, this make and model)		

### Co-Pilot Information

<b>Certificate:</b>	Airline Transport; Flight Instructor; Commercial	<b>Age:</b>	30
<b>Airplane Rating(s):</b>	Multi-engine Land; Single-engine Land	<b>Instrument Rating(s):</b>	Airplane
<b>Other Aircraft Rating(s):</b>	None	<b>Instructor Rating(s):</b>	Airplane Single-engine; Instrument Airplane
<b>Flight Time:</b>	4800 hours (Total, all aircraft), 1000 hours (Total, this make and model)		

### Aircraft and Owner/Operator Information

<b>Aircraft Make:</b>	Airbus Industrie	<b>Registration:</b>	N814AW
<b>Model/Series:</b>	A319-132	<b>Engines:</b>	2 Turbo Fan
<b>Operator:</b>	America West Airlines Inc.	<b>Engine Manufacturer:</b>	International Aero Engines
<b>Operating Certificate(s) Held:</b>	Flag carrier (121)	<b>Engine Model/Series:</b>	IAE-2524-A5
<b>Flight Conducted Under:</b>	Part 121: Air Carrier - Scheduled		

### Meteorological Information and Flight Plan

<b>Conditions at Accident Site:</b>	Visual Conditions	<b>Condition of Light:</b>	Day
<b>Observation Facility, Elevation:</b>	MAF, 2871 ft msl	<b>Weather Information Source:</b>	Weather Observation Facility
<b>Lowest Ceiling:</b>	None	<b>Wind Speed/Gusts, Direction:</b>	8 knots / , 230°
<b>Temperature:</b>	17° C	<b>Visibility</b>	10 Miles
<b>Precipitation and Obscuration:</b>			
<b>Departure Point:</b>	Houston, TX (IAH)	<b>Destination:</b>	Phoenix, AZ (PHX)

## Airport Information

<b>Airport:</b>	Midland International (MAF)	<b>Runway Surface Type:</b>	Asphalt
<b>Runway Used:</b>	16R	<b>Runway Surface Condition:</b>	Dry
<b>Runway Length/Width:</b>	9501 ft / 150 ft		

## Wreckage and Impact Information

<b>Crew Injuries:</b>	5 None	<b>Aircraft Damage:</b>	None
<b>Passenger Injuries:</b>	84 None	<b>Aircraft Fire:</b>	None
<b>Ground Injuries:</b>	N/A	<b>Aircraft Explosion:</b>	None
<b>Latitude, Longitude:</b>	31.942500, -102.201944		

## Administrative Information

<b>Investigator In Charge (IIC):</b>	Jason A Ragogna	<b>Adopted Date:</b>	12/03/2004
<b>Investigation Docket:</b>	NTSB accident and incident dockets serve as permanent archival information for the NTSB's investigations. Dockets released prior to June 1, 2009 are publicly available from the NTSB's Record Management Division at <a href="mailto:pubinq@ntsb.gov">pubinq@ntsb.gov</a> , or at 800-877-6799. Dockets released after this date are available at <a href="http://dms.nts.gov/pubdms/">http://dms.nts.gov/pubdms/</a> .		

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